Temasek Defence Systems Institute

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Performance & Diagnostics Analysis for A Civil Aero Gas Turbine Engine

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Aim:

To investigate the degradation of the core gas path components of a civil aero turbofan engine using **GPA**

Methodology:

Refinement and

Results:

- DP & OD adaptation errors well within 1% ullet
- Top 9 set selected as optimal diagnostic measurement set 20 ulletof 21 fault cases validated
- Comp2 definitely degraded based on 3 likely fault cases
- Comp2 Turb2 most likely 2 component degraded fault case (fouling) due to realistic health parameter predictions and trend overtime



Maintenance action probably carried out between 185 and 500 hours leading to loss of GPA prediction accuracy overtime

Research Benefits:

- Reducing gas turbine downtime and improving availability
- Reducing unnecessary maintenance costs in overhauling gas • turbines at OEM predetermined intervals

Future Work:

- Integrate together with other techniques to deal better with noise and produce better degradation predictions
- Continue with prognostic analysis to determine remaining useful • life



Turb 3 Turb 1 Turb 2 Turb 3 Turb 1 Turb 2 Turb 3 Comp 2 Comp 3 Comp 3 Turb 2 Turb 2 Turb 1 Component Fault Cases

■ 185 hr ■ 500 hr ■ 825 hr ■ 870 hr ■ 1140 hr ■ 1300 hr ■ 1450 hr

Initial Diagnostic Result with Raw Data

Diagnostic Result with Postprocessed Data



185 Hrs

500 Hrs

825 Hrs

1300 Hrs

Urban, L. A. (1972) 'Gas-path Analysis Applied to Turbine Engine Condition Monitoring', in AIAA/SAE 8th Joint Propulsion Specialist Conference. 1.

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- Li, Y. G., Ghafir, M. F. A., Wang, L., Singh, R., Huang, K. and Feng, X. (2011) 'Nonlinear Multiple Points Gas Turbine Off-Design Performance Adaptation Using a 3. Genetic Algorithm', Journal of Engineering for Gas Turbines and Power, 133(7), pp. 071701-071701-9.

